PROJECT: SAMS
ASS'T NUMERICATURE: SERVO POWER AMPLITIER SYSTEM: FLECTRICAL SUBSYSTEM
ASS'T P/N: 5172071177 SHEET:

IMEA AEF.	THEA BEV.	MAME DIT B DRAVING REF. DESIGNATION	FAILURE MODE AND CAUSE	FAILURE EFFECT ON END LIEN	HDWR / FUNC. 2/1R CRITICALITY	RATIONALE FOR ACCEPTANCE
2620	* M	PHASE LOCK LOOP 01T-6 SCHEMATIC 2563729	MODE: LOSS OF OR PHICORRECT TACHOMETER FEED BACK TO SERVO LOOP AND MCIU. CAUSE(S): (1) FAILURE OF PULSE SHAPER. (2) FILTER FAILS OPEN. (3) VCO-PLL FAILURE OF 2**6 DIVIDER	COUNTER PRODUCING RATE DATA IS ERROHEOUSLY OR HOI UPDATED. MAT RESULT IN A MAKEMUN REGATIVE (CCU) COMMAND. MOTOR WILL BE DRIVEN AT RAXIMUM RATE CCV. RATE DATA TO THE RETUYGPC VILL BE ERROHEOUS. AUTO BRAKES INTITATED. LOSS OF LIMPING DURING END EFFECTOR CAPTURE. WORST CASE UNEXPECTED MOTION. JOINT RUMAWAY. AUTOBRAKES. REDUNDANT PATHS REMAINING AUTOBRAKES	DESIGN FEATURE COMPARATORS AN INVEGRATED CIR APPLICATION CO SPAR-RMS-PA.OO ALE RESISTORS FROM ESTABLISH INCREASED BY E DERAFED IN ACC ELECTROLYTIC C RADTOGRAPHIC I THE DESIGN UTI IMPLEMENTED US CMOS DEVICES O SIGNIFICANT OPP DEVICE RELIABI ARE ADDITIONAL PARAMETERS IN HANDLING PRECA	ID OPERATIONAL AMPLIFIERS ARE STANDARD LINEAR CUITS WITH MATURE MANUFACTURING TECHNOLOGY. MISTRATH'S ARE IN ACCORDANCE WITH 13. AND CAPACITORS USED IN THE DESIGN ARE SELECTED ED RELIABILITY (ER) TYPES. LIFE EMPECTANCY IS NESURING THAT ALL ALLOWABLE STRESS LEVELS ARE ORDANCE WITH SPAR-RNS-PA.003. ALL CERAMIC AND APACITORS ARE ROUTINELY SUBJECTED TO MSPECTION. LIZES PROVEN CIRCUIT TECHNIQUES AND IS ING CMOS LOGIC DEVICES. PERATE AT LOW POWER AND HENCE DO NOT EXPERIENCE ERATING STRESSES. THE TECHNOLOGY IS MATURE, AND LITY HISTORY IS WELL DOCUMENTED, ALL STRESSES LY REDUCED BY DERAFING THE APPROPRIATE ACCORDANCE WITH SPAR-RNS-PA.003. SPECIAL UTIONS ARE USED AT ALL STAGES OF MANUFACTURE TO EXSTRESS DUE TO ELECTROSTATIC DISCHARGE. MPOMENTS ARE SCREENED TO THE REQUIREMENTS OF 3.
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THE A BRY DEALINE WITE BRANCH OFF. CAUSE 1 PRACT OCT ON OTHER CONTROL CAUSE 1 PRACT OCT ON OTHER CONTROL CAUSE 1 PRACT OCT ON OTHER CONTROL CAUSE 2 PRACT OCT ON OTHER CONTROL CAUSE 2 PRACT OCT ON OTHER CONTROL CAUSE 2 PRACT OCT ON OTHER CONTROL CAUSE OF PRODUCING MARE CAUSE OF PRODUCING MARK CAUSE OF PRODUCING	RITICAL IT	EHO LIOT		IDJECT: SAMS SS'Y NOMERCEATURE: S	THE SYSTEM SYSTEM ASS'Y P/N: 51140F1177	SHEET:
SCHMANIC INCORRECT INCORR		DRAWING BEF.	AND	ON	2/18	
REPARED BY: MFMQ SUPERCEDING DATE: 11 SEP 86 APPROVED BY: DATE: 24 JUL 91 CH BEV	2620 1	LOOP GIT-6 SCHEMATIC	LOSS OF OR INCORRECT TACHOMETER FEED BACK TO SERVO LOOP AND MCIU. CAUSE(S): (1) FAILURE OF PULSE SHAPER. (2) FILTER FAILS OPEN. (3) VCO-PLL FAILURE OF 2**6 DIVIDER	PRODUCING RATE DATA 1S ERROWEGUSLY OR HOT UPDATED. MAY RESULT IN A MAXIMUM HEGATIVE (CCW) COMMAND. MOTOR WILL BE DRIVEN AT MAXIMUM RATE CCW. RATE DATA TO THE MCTU/CPC WILL BE ERROWEGUS. AUTO BRAKES INITIATED. LOSS OF LIMPING DURING END EFFECTOR CAPTURE. WORST CASE UNEXPECTED MOTION, JOINT RUNAWAY. AUTOBRAKES. REDUMDANT PATHS REHAINING AUTOBRAKES	ACCEPTANCE 18519 THE SPA 18 SUBJECTED TO THE FOLLOWING ENVIRONMENTAL TESTIAN SRU. O VIBRATION: LEVEL AND DUNATION - REFERENCE TABLE 4 O THERMAL: PLUS 70 DEGREES C TO -25 DEGREES C DUNAFIO 1 1/2 CYCLES THE SPA 18 THEM TESTED AS PART OF THE JOINTS ACCEPTANCE T (VIBRATION AND THERMAL VACUUM TEST). THE SPA'S/JOINTS UNDERGO AND SYSTEM TESTS (TPS18 RMS STRONGBACK AND TPS52 FLAT FLOOR TESTS) WHICH VERIFIES THE ABSENCE OF THE FAILURE MODE. QUALIFICATION TESTS THE SPA 18 SUBJECTED TO THE FOLLOWING SRU QUALIFICATION 1 ENVIRONMENTS. THE SPA WAS ALSO TESTED AS PART OF THE JOIN QUALIFICATION TESTS. O VIBRATION: LEVEL AND DURATION - REFERENCE TABLE 4 O SHOCK: 200/11 MS/3 AXES (6 DIRECTIONS) D THERMAL VAC: *81 DEGREES C TO -36 DEGREES C (6 CYCLES) 1X10°-6 FORC O HUMIDITY: TESTED WITH THE SHOULDER JOINT O ENC: MIL-STD-461 AS MODIFIED BY SL-E-DOO2 (TESCE), CEO1, CEO3, CSO1, CSO2, CSO6, REO1, REO2 (M/B), RSO1) FLIGHT CHECKOUT PDRS OPS CHECKLIST (ALL VEHICLES) JSC 16987	ESTS

PROJECT: SAMS

ASS'T MINITACEATURE: STAVO POUCH ANDLIFIER ASS'Y 17/N: 51120F1177 SHEFT: 3

PMEA REF.	IMEA REV.	DESIGNATION	FAILURE MUDE AND CAUSE	FAILUME EFFECT ON END LIEN	HOUR / FUNC. RATIONALE FOR ACCEPTANCE 2/IR CRITICALITY SCREENS: A-PASS, B-PASS, C-PASS
PREPARED B	1 H	PHAST LOCE LOOP OTT-6 SCHEMATIC 2363723	NUDE: LOSS OF ON INCORRECT TACHOMETER STED BACK TO SERVO LOUP AND MCIU. CAUSE(S): (1) FAILURE OF PULSE SHAPER. (2) FILTER FAILS OPEN. (3) VCO-PLL FAILURE OF 2**6 DIVIDER	COUNTER PRODUCING RATE DATA IS E RROWEOUSLY OR ROT WIDATED. MAT RESULT IN A MAXIMUM MEGATIVE (CCW) COMMAND. MOTOR WILL BE DRIVEN AT MAXIMUM RATE CCW. RATE DATA TO THE MCTU/GPC WILL BE ERROWEOUS. AUTO BRAKES ENTITIATED. LOSS OF LIMPING DURING END EFFECTOR CAPTURE. WORST CASE UNEXPECTED MGTION. JOINT RUNAWAT. AUTOBRAKES. REDUNDANT PATHS REMAINING AUTOBRAKES	UNITS ARE MANUFACIURED UNDER DOCUMENTED QUALITY CONTROLS. INESE COMINGUS ARE EXERCISED THROUGHOUT DESIGN PROCURTREMS, PLANWING, RECEIVING, PROCESSING, FABRICATION, ASSEMBLY, TESTING AND SHIPPING OF THE UNITS. MANDATORY INSPECTION POINTS ARE EMPLOYED AT VARIOUS STAGES OF FABRICATION ASSEMBLY AND TEST, GOVERNMENT SOURCE INSPECTION IS INVOKED AT VARIOUS CONTROL LEVELS. EEE PARTS INSPECTION IS PERFORMED AS REQUIRED BY SPAR-MRS-PA, DOS, EACH SEE PART IS QUALIFIED AT THE PART LEVEL TO THE REQUIREMENTS OF THE APPLICABLE SPECIFICATION. ALL EEE PARTS ARE 100% SCREENED AND BURNED IN, AS A MINIMM, AS REQUIRED BY SPAR-MS-PA, DOS, BY THE SUPPLIER. ADDITIONALLY, EEE PARTS ARE 100% RE-SCREENED IN ACCORDANCE WITH REQUIREMENTS, BY AN INDEPENDENT SPAR APPROVED TESTING PACIFITY. DPA IS PERFORMED AS REQUIRED BY PA, DOS ON A RANDOMLY SELECTED SX OF PARTS, MAINTAMS PIECES INFORMALLY, EEE PARTS ARE 100% RE-SCREENED IN PIECES FOR EACH LOT MUMBER/DATE CODE OF PARTS RELEVED. WIRE IS PROCURED TO SPECIFICATION MIL-M-22759 OR MIL-M-BISSI AND INSPECTED AND TESTED TO MASA JSCHOOLD STANDARD MUMBER 95A. RECEIVING INSPECTED WERSTED TO MASA JSCHOOLD STANDARD MUMBER 95A. RECEIVING INSPECTED WERSTED TO MASA JSCHOOLD STANDARD MUMBER 95A. RECEIVING INSPECTED PARTS RELEVED. PRINTED COLUMENT PROVIDE ADEQUATE TRACEABILITY INFORMATION AND SCREENING DATA CLEARLY IDENTIFIES ACCEPTABLE PARTS. PARTS ARE INSPECTED THROUGHOUT MANUFACTURE AND ASSEMBLY AS APPROPRIATE TO THE MANUFACTURING STAGE COMPLETED. THESE INSPECTIONS INCLUDE, PRINTED CIRCUIT BOARD INSPECTION FOR CORRECT SOLDERING, MIRE LOOPING, STRAPPING ETC. OPERATORS AND INSPECTION POINT) P.C. BD. INSTALLATION INSPECTION FOR ADEQUATE PROCESSING IS PERFORMED USING ULTRAVSOLET LEGHT TECHNIQUES. POST P.C. BD. INSTALLATION INSPECTION, CHECK FOR CONRECT BOARD INSTALLATION, ALIGNMENT OF BOARDS, PROPER CONNECTOR CONTACT MATHS AND ASSEMBLY AS APPROVED TO THE SECTION, WORKMANSHIP AND CLEANLINESS COMPARANSHIP (SPAR/GOVERNMENT REP. MANDATORY INSPECTION POINT). PRE-ACCEPTANCE TEST INSPECTION, WHICH INCLU
	_				APPROVED BY: DATE: 24 JUL 91 CIL REV: _1

FROJECT: SEMS ASS-T MOMENCLATURE: SERVO POWER AMPLIFIER MATIONALE FOR ACCEPTANCE THEA THEA HAME, DEF, & FAILURE MEDE FATTURE EFFECT HOUR / FUNC.

rmtå Rt1.	PHT A REV.	DESIGNATION DESIGNATION HAMP OFF B	FAILURE MEDE AND CAUSE	FAILURE EFFECT CM END LITER	NOWN / FUNC. RATIONALE FOR ACCEPTANCE 2/18 CRITICALITY SCREEKS: A-PASS, B-PASS, C-PASS
7670		PMASF LOFE LOWP 017 6 SCHEMATIC 2563723	MTDE: LOSS OF OR INCORRECT TACHCHETER FEED BACK TO SIRVO LOWP AND MCTU. CAUSE(\$): (1) FAILURE OF PULSE SHAPER. (2) FFLTER FATLS OPEN. (3) VCO-PLL FAILURE OF 2**6 DIVIDER	COMITATE PRODUCING BATE PRODUCING BATE PRODUCING BATE OR NOT UPDATED. MAT RESULT IN A MAXIMUM RATE COMMAND. MOIOR WILL BE DRIVEN AT MAXIMUM RATE CCV. RATE DATA TO THE MCTU/GPC WILL BE ERRONEOUS. AUTO BRAKES INTITATED. LOSS OF LIMPING END EFFECIOR CAPTURE. WORST CASE UNEXPECTED MOTION. JOINT RUMANAT. AUTOBRAKES. REDUNDANT PATHS REMAINING. AUTOBRAKES	A TEST READINESS REVIEW (TRR) WHICH INCLUDES VERIFICATION OF TEST PERSONNEL TEST DOCUMENTS, LEST GOUTEMENT CALIBRATION, VALIDATION SIATUS AND HARDWARE CONTIGURATION IS CONVENED BY OUALITY ASSUMANCE IN COMJUNCTION UNITH REGISTER AS APPLICABLE, AND THE GOVERNENT REPRESENTATIVE, PRIOR TO THE START OF ANY FORMAL TESTING (ACCEPTANCE OR GUALIFFICATION). ACCEPTANCE TESTING (ATP) INCLUDES AMBIENT PERFORMANCE, THERMAL AND VIBRATION TESTING, (SPAR/GOVERNMENT REP. INTEGRATION OF UNIT TO JOINT SRU - INSPECTIONS INCLUDE CROUNDING CHECKS, COMMECTORS FOR BENT OR PUSHBACK CONTACTS, VISUAL, CLEANLINESS, INTERCONNECT WIRTH AND POWER UP TEST TO THE APPROPRIATE JOINT INSPECTION TEST PROCEDURE (ITP) ETC. JOINT LEVEL PRE-ACCEPTANCE TEST INSPECTION, INCLUDES AN AUDIT OF LOWER TIRE INSPECTION COMPLETION, AS BUILT CONFIGURATION VERTIFICATION TO AS DESIGN ETC. JOINT LEVEL ACCEPTANCE TESTING (ATP) INCLUDES AMBINET, VIBRATION AND THERMAL-VAC TESTING. (SPAR/GOVERNMENT REP MANDATORY INSPECTION POINT). SAMS SYSTEMS INTEGRATION, THE INTEGRATION OF MECHANICAL ARM SUBASSEMBLES AND THE FLIGHT CABIN COULPMENT TO FORM THE STANS INSPECTION ARE PERFORMED AT EACH PHASE OF INTEGRATION WHICH HICKURES GROUNDING CHECKS, THRU WIRING CHECKS, WIRING ROUTHING, INTERFACE CONNECTORS FOR SHIT OR PUSH BACK CONTACTS ETC. SAMS SYSTEMS IESTING - STRONGBACK AND FLAT FLOOR AMBIENT PERFORMANCE TESTING - STRONGBACK AND FLAT FLOOR AMBIENT.
I		MELINO	SUPERCEDING D	ATE: 11 SEP 86	APPROVED BY: DATE: 24 JUL 91 CIL REY: _

SUPERCEDING DATE: 11 SEP 86 HING PREPARED BY

KITIC	L ITE	HO LIOT	P) A:	ROJECT: SRMS SS+V NO+! (ATURE)	SERVO POUER AMPLITTER	SYSIEM: ELECTRICAL SUBSYSTEM ASSY P/N: 51120F1177 SHEEL
THEA REF.	PHEA REV.	NAME DIY & DRAWING REF. DESIGNATION	FATLURE MODE AND FAILSE	FAILURE EFFECT ON END ITEM	HDWR / FUNC. 2/18 CRITICALITY	MATIONALE FOR ACCEPTANCE
2620		PHASE LOCK LOOP GIT-6 SCHEMATIC 2563723	MUNIT: LOSS OF OR INCORRECT TACHOMETER FEED BACK TO SERVO LOOP AND MCTU. CAUSE(S): (1) FAILURE OF PULSE SHAPER. (2) FILTER FAILS OPEN. (3) VCO-PLL FAILURE OF Z**6 DIVIDER	COUNTER PRODUCING RATE DATA IS ERROMEOUSLY DR NOT UPDATED. MAY RESULT IN A MAXIMUM MEGATIVE (CCU). COMMAND. MOTOR WILL BE DRIVEN AT MAXIMUM RATE CCW. RATE DATA TO THE MCTU/GPC WILL BE ERROMEOUS. AUTO BRAKES INTITATED. LOSS OF LIMPING DUNING END EFFECTIOR CAPTURE. WORST CASE UMEXPECTED MOTIOM. JOINT RUNAWAY. AUTOBRAKES. REDUNDANT PATHS REMAINING AUTOBRAKES	FAILURE HISTOR THE FOLLOWING FAR 3320: S/H 313 OCT DESCRIPTION FILTER OUTPUT	FAILURE AMALYSIS REPORT(S) ARE RELEVANT: 82 VOLTAGES ONE - HALF EXPECTED VALUES, FOUND (700), SUSPECTED ELETRO-STATIC DISCHARGE.
PAGED BY:	<u>H11</u>	ug	SUPERCEDING DATE	: 11 SEP AA	APPENVEN BY	DATE: 24 JUL 91 CIL REV

PROJECT: SAMS
ASS'Y MIMERICATURE: SERVO POUER AMPLIFIER
ASS'Y PAR: 511401177 SIM11: 6

PROFES OF A SCHMANTC	THEA REF.	THEA REV.	DRAVING REF. DESIGNATION	FAITURE MODE AND CAUSE	FATLURE EFFECT ON END ETEM	HOUM / FUNC. RATIONALE FOR ACCEPTANCE 2/IR CRITICALITY SCREENS: A-PASS, B-PASS
	2620		PHASE LOCK LOOP Q1T-6 SCHEMATIC	MODE: LDSS OF OR INCORRECT TACHOMETER FEED BACK TO SERVO LOOP AND MCTU. CAUSE(\$): (1) FAILURE OF PULSE SHAPER. (2) FILTER FAILS OPEN. (3) VCO-PIL FAILURE OF	COUNTER PRODUCING RATE DATA IS ERRONEOUSLY OR NOT UPDATED. MAY RESULT IN A MAKINUM NEGATIVE (CCW) COMMAND. MOTOR WILL BE DRIVEN AT MAXIMUM RATE CCW. RATE DATA TO THE MCIU/GPC WILL BE ERROMEOUS. AUTO BRAXES INTITATED. LOSS OF LIMPING DURING END EFFECTOR CAPTURE. WORST CASE UMEXPECTED MOTION, JOINT RUNAMAY. AUTOBRAKES. REDUNDANT PATHS REMAINING	OPERATIONAL EFFECTS JOINT RUMANAY, AUTOBRAKES, CANNOT USE COMPUTER SUPPORTED MODES, DIRECT DRIVE AND BACKUP AVAILABLE, ARM WILL NOT STOP AUTOMATICALLY IF AN UNDETECTED FAILURE OF THE AUTO BRAKES SYSTEM MAS PREVIOUSLY OCCURRED BRAKES CAN BE APPLIED MANUALLY. CREW ACTION THE CREW WILL BE TRAINED TO ALWAYS OBSERVE WHETHER THE ARM IS RESPONDING PROPERLY TO COMMANDS. IF IT ISN'T, APPLY BRAKES. MISSION COMSTRAINT OPERATE UNDER VERNIER RATES WITHIN TO FT OF STRUCTURE. THE OPERATOR MUST BE ABLE TO DEVECT THAT THE ARM/PAYLOAD IS RESPONDING PROPERLY TO COMMANDS VIA WINDOW AND/OR CCTV VIEWS DURING ALL ARM OPERATIONS. OMRSD OFFLINE VERIFY ABSENCE OF TACHOMETER BITE ON ABE DATA. OMRSD ONLINE INSTALLATION NOME OMRSD ONLINE TURNAROUND